

# SPACE

Towards sensing and perception for autonomous berthing – force-torque sensor and an algorithm for the gripping point pose estimation.

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### Key technologies for orbital and planetary robotics

- LARIS Gripper
- LARIS Docking System
- Standard Gripping Fixture
- Robotic Arm TITAN
- Force and Torque Sensor
- Vision System

### Context



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# **PIAP Space Force and Torque Sensor**



- Redundant strain gauges and measuring channel fail safe
- Custom-designed redundant electronic controller, fitted inside the body of the sensor – path to fligth
- Temperature monitoring and control accuracy and reliability
- Calibration method wide range of measurments

## **FTS Parameters**



- Measurement range of 50 N and 50 Nm,
- Overload capacity of 300%,
- Accuracy ≤3% (worst case)
- Easy adaptation to different measurement ranges.
- Operating range of -40/+100°C (survival range -100/+125°C)
- Φ125mm, 70mm height
- <1 kg
- More parameters in the paper!

# **FTS** Principle of Operations

The sensor converts force and torque input F to signal output U. The sensor measures three components of forces and three components of torques in  $\mathbb{R}3$ .

$$\boldsymbol{U} \approx f(\boldsymbol{F})$$

$$F = \begin{bmatrix} F_x & F_y & F_z & M_x & M_y & M_z \end{bmatrix}^T$$

$$\boldsymbol{U} = \begin{bmatrix} U_1 & U_2 & \dots & U_n \end{bmatrix}^T$$

Each signal output **U** is affected by each component of force and torque **F**, as **no sensor is perfectly decoupled**. This is shown by the coupling matrix C. The output signal is assumed to be proportional to applied load, as the sensor is designed to operate in range of elastic deformations.

 $U \approx CF$ 

$$\boldsymbol{C} = \begin{bmatrix} C_{11} & \cdots & C_{16} \\ \vdots & \ddots & \vdots \\ C_{n1} & \cdots & C_{n6} \end{bmatrix}$$

Output signal is also affected by a **bias E** that was not fully compensated. This offset is assumed to be constant in whole measurement range and different for each output.

$$\boldsymbol{U} = \boldsymbol{C}\boldsymbol{F} + \boldsymbol{E}$$
$$\boldsymbol{E} = \begin{bmatrix} E_1 & E_2 & \dots & E_n \end{bmatrix}^T$$

A simple transformation is done to estimate forces and torgues acting upon sensor. This operation can be employed in order to use the force and torgue sensor as a measurement instrument.

$$F = C^{-1}(U - E)$$

# Mechanical structure and strain gauges

- Stewart platform like internal structure,
  - low non-linearity,
  - high accuracy
  - advantageous ratio of stiffness to mass.
- Redundant measurement channels,
  - vacuum compatible
  - extended temperature range.
- Both sets of strain gauge bridges are connected to individual, independent parts of controller with separate communication and power connections.



### **PIAP Space Controller**

- Stacked PCBs fitted inside the FTS
  - prepared to use space rated components.
- Redundant architecture with two microcontrollers
- Software thermal compensation
- Sensor's thermal control
  - temperature sensors redout and power heaters control



### **Controller Software**

- Proprietary software for driving the controller, permitting data acquisition and visualization,
- CAN bus for communication and operates on Windows and Linux (Ubuntu 18.04) operating systems
- CANopen



# **Calibration Station**



- Applying loads simultaneously in multiple axes,
- Compatible with climate chamber for calibrations at -40 and +40,
- Equipped with precisely machined weights, allow for application of loads from 15 N or 7,5 Nm up to 800 N or 400 Nm
- Calibrated dynamometers

• Vacuum compatible Calibration Stations with much wider temperature range is planned.

### **Calibration Results**



Output U in function of positive force component Fx



- Full calibrations at temperatures -20, 0, 20 and 40°C
- Coupled but linear behavior

### **Tests Results**



- Ageing in the range from -40 to +100 °C to reduce hysteresis of zero signals.
- Test in the full range of from -50 to 50°C.
- TVAC tests are planned in upcoming months

## **FTS Conclusions and Timeline**

- **Q2 2023** DM-1 off-the-shelf Force and Torque Sensor model calibrated and tested,
- Q3 2023 Custom electronics tested with DM-1 and DM-2 models
- Ongoing tests and calibration of the customized DM-2 model with redundant channels
  - TVAC and vibration tests till end of the 2023 TRL 6.

### Calibration method validated!

• The proposed calibration may be used for any kind of force and torque sensor with linear output and with any number of outputs

### • TVAC calibration is planned to start in 2024.



# **Vision System**

- Vision system for estimation of LAR pose in 6 DOF without markers in close distance 0.2 – 1.2m
- LAR diameter and height has to be known
- Sensor to be mounted on robotic endeffector -> low mass & small envelope
- Linear accuracies:
  - 20mm@z=1,2m;
  - 10mm@z=500;
  - 4mm@z=250mm
- Angular accuracies:
  - 1deg@z=1,2m;
  - 0.5deg@z=500;
  - 0.2deg@z=250mm



# **Vision System Hardware**

#### Hardware agnostic, but:



### CubeEye/Meere Company S100D

- Indirect (phase-shift) ToF, 940nm
- Operating range: 0.2 4m
- FoV: 60x45 deg
- Resolution: 640x480
- Frame rate: max 30fps
- Not tested regarding space environment



### Aetina AIE-CT41 Processing unit

• NVIDIA Jetson TX2 NX

# Vision System Algorithm



- Using open-source PCL library
- Language: c++
- Algorithm can be used on any point cloud map
- Algorithm implemented in two application:
  - As EGSE (for parameters adjustment and inspection)
  - As module on Robotic arm (position feedback to motion controller)
- Additional filtering of results to correct stability

# Vision System Software

### SW-1

- GUI written on QT5
- Fine tuning of all algorithm and camera parameters
- Used mainly to verify algorithm behavior
- Works with saved point cloud data and with live camera data,

#### SW-2

- 2 modes (with/without visualization)
- **CAN-Open** implemented for communication with Motion Controller
- Optimized for performance
- Possibility to execute commands by terminal or GUI (in mode with visualization)



### Test set-up

- 3x3x3m aluminium cage covered with material absorbing IR radiation
- Printed version of LAR from Sentinel-3
- LAR mounted on flat Surface covered by Mylar + Kapton
- Additional 6 different MLI material/components ready to test
- Handheld CMM used for accuracy measurements
- Camera mounted on rails (linear movement) and tripod head (angular movement)
- Different lightning conditions



## **VS Verification**

Verification procedure:

- Set camera position on rails and orientation on tripod head
- Perform batch of VS measurement
- Perform CMM measurement of the position of the LAR grasping points in the camera reference frame
- Compare results
- Repeat for different position on z axis



### **Test results I**

- Accuracy stable over distance
- X and Z accuracy below 6mm at whole range
- Accuracy of x and z ~1mm @z=250mm
- Accuracy on y axis outside required accuracy

   in practice it's least significant
- Range currently limited from 250mm
- At z=250mm big instabilities
- Angular measurement repeatability not affected by distance
- Poor angular accuracy on close distance due to smaller dataset and higher noise
- VS performance supposed to be better with black Kapton MLI



### **Test results III**



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## **VS Summary**

- VS works well enough to provide accurate movement for LAR grasping
- Calculation frequency is ~30Hz
- Algorithm can be further improved for better stability and accuracy
- Different MLI materials to be tested
- Plan to check ToF camera with wider FoV to increase repeatability in close range
- VS will be used during the TITAN tests







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# **BE OUR COMMON SUCCESS**

We look forward to hearing from you

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